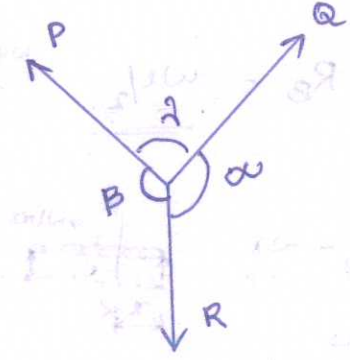


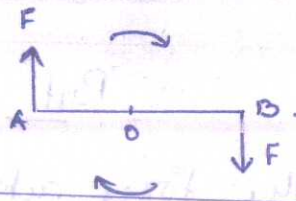
Revision : 2015

Course title : THEROY OF STRUCTURES-I

Qst. No	Scoring indicators	Split Up Marks	Sub Total	Total
I	<p style="text-align: center;"><u>Part - A.</u></p> <p>1. IF the Forces acting on a particle keep in equilibrium then each force is proportional to the sine of the angle between the two other two.</p> <div style="border: 1px solid black; padding: 5px; display: inline-block; margin: 10px;"> $\frac{P}{\sin \alpha} = \frac{Q}{\sin \beta} = \frac{R}{\sin \gamma}$ </div> 		2	10
	<p>2. With in the elastic limit the stress is directly proportional to strain.</p> <p style="text-align: center;">Stress \propto strain:</p> <div style="border: 1px solid black; padding: 5px; display: inline-block; margin: 10px;"> $\sigma = E \cdot \epsilon$ </div>		2	
	<p>3. In a bending beams, a point is known as a point of contra flexure if it is a location where bending moment is zero (changes its sign).</p>		2	
	<p>4. It is the line on the neutral plane on the beam cross-section. All fibres situated above the neutral axis are under compression and those below it are under tension.</p>		2	

The line or axis of stress is zero. This line is called neutral axis.

5. Two equal, unlike, parallel forces whose lines of action are different from a couple.



2

Part B

II

1. As the beam is symmetric.

$$R_A = R_B = \frac{wl}{2}$$

SFD

$$SF_{xx} = \frac{wl}{2} - wx$$

$$SFA = \frac{wl}{2}$$

$$SFB = -\frac{wl}{2}$$

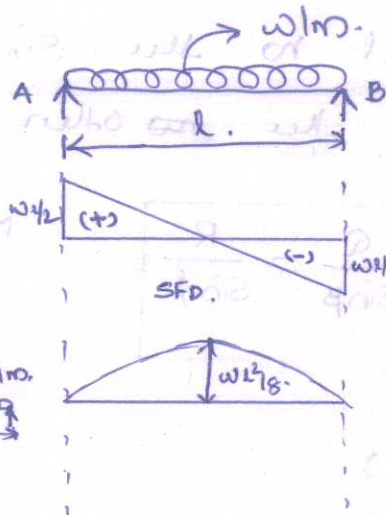
BMD

$$M_{xx} = \frac{wl}{2} \cdot x - \frac{wx^2}{2}$$

$$M_A = 0, \quad M_B = \frac{wl^2}{2} - \frac{wl^2}{2} = 0$$

Moment at Centre (max. Bm.)

$$M_{max} = \frac{wl^2}{8}$$



6

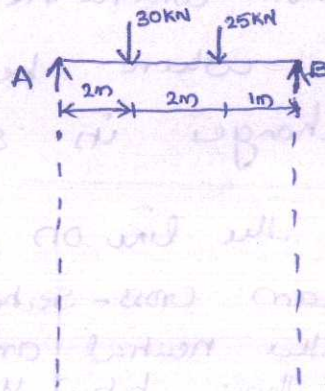
2. $R_A + R_B = 55 \text{ kN}$

$$M_A \Rightarrow R_B \times 5 = (25 \times 7) + (30 \times 2)$$

$$5R_B = 100 + 60$$

$$R_B = \frac{160}{5} = \underline{32 \text{ kN}}$$

$$R_A = 55 - 32 = \underline{23 \text{ kN}}$$



6

$$3.) \quad t = T_2 - T_1 = 80^\circ\text{C}$$

$$L = 3\text{m}$$

$$T_1 = 10^\circ\text{C}$$

$$T_2 = 90^\circ\text{C}$$

Stress

$$\sigma = \left(\frac{\alpha t L - \beta}{L} \right) \times E$$

$$E = 2 \times 10^5 \text{ N/mm}^2$$

$$\alpha = 12 \times 10^{-6} / ^\circ\text{C}$$

$$\beta = 1\text{mm}$$

$$= \left(\frac{12 \times 10^{-6} \times 80 \times 3000 - 1}{3000} \right) \times 2 \times 10^5$$

$$= \underline{\underline{125.33 \text{ N/mm}^2}}$$

4)

$$E = 2N(1+\mu)$$

$$E = 1.2 \times 10^5 \text{ N/mm}^2$$

$$N = 4.8 \times 10^4 \text{ N/mm}^2$$

$$\mu = \frac{E}{2N} - 1 = \frac{1.2 \times 10^5}{2 \times 4.8 \times 10^4} - 1$$

$$\mu = \underline{\underline{0.25}}$$

$$E = 3K(1-2\mu)$$

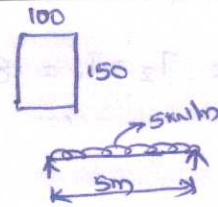
$$K = \frac{E}{3(1-2\mu)} = \frac{1.2 \times 10^5}{3(1-2 \times 0.25)}$$

$$= \underline{\underline{8 \times 10^4 \text{ N/mm}^2}}$$

5)

- The material of the shaft is uniform throughout
- The twist along the shaft is uniform.
- The shaft is of uniform cross-sections throughout
- Cross sections of the shaft which are plane before twist remain plane after twist.
- All radii which are straight before twist remain straight after twist.

6) $w = 5 \text{ kN/m} = 5 \text{ N/mm}$
 $L = 5 \text{ m}$
 $b = 100 \text{ mm}$ $d = 150 \text{ mm}$.



$$M = \frac{wl^2}{8} = \frac{5 \cdot 625}{8} = 15.625 \text{ kNm} = 15.625 \times 10^6 \text{ Nmm}$$

$$I = \frac{bd^3}{12} = \frac{100 \times 150^3}{12} = 42187500 \text{ mm}^4$$

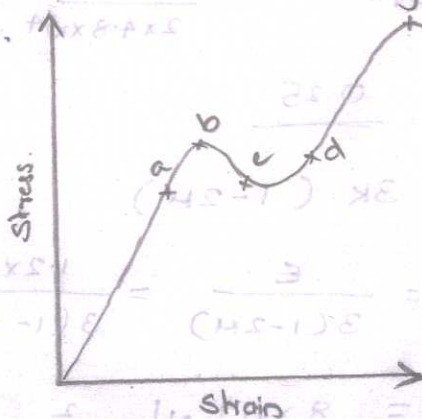
$$y_{\text{max}} = d/2 = 150/2 = 75 \text{ mm}$$

From Bending eqn.

$$\frac{M}{I} = \frac{\sigma_{\text{max}}}{y_{\text{max}}} \Rightarrow \sigma_{\text{max}} = \frac{M \times y_{\text{max}}}{I}$$

$$= \frac{15.625 \times 75 \times 10^6}{42187500} = 27.77 \text{ N/mm}^2$$

- 7) a → limit of proportionality.
 b → elastic limit
 c → upper yield point.
 d → lower yield point.
 U → ultimate stress load
 F → Fracture stress load.



- Limit of proportionality :- It is a load upto which stress strain relation is linear and we get straight line on graph.
- Elastic limit (b) :- It is the load upto which material retains elasticity. It may regain its original size if load is removed. It is the point beyond which some permanent deformation may occur.
- Upper yield point (c) :- It is the point at which material gives and yield stress is the lowest stress at which material is appreciably deformed without the load.

(8)
6

30

6

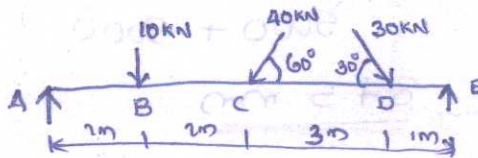
- lower yield point :- It is the point from which material enters ductile.
- ultimate stress (U) :- It is the stress developed at maximum load. In brittle materials it corresponds to stress at breaking point.
- Fracture load (F) :- It is the load at which the material gets fractured the stage during U and F the specimen enters a fracture zone and neck formation takes place.

Part - c

III

a) $40 \sin 60^\circ = 34.64 \text{ kN}$

$30 \sin 30^\circ = 15 \text{ kN}$



$\sum V = 0$

$R_A + R_E = 10 + 34.64 + 15$

$R_A + R_E = \underline{59.64 \text{ kN}}$

$\sum M_A = 0$

$(10 \times 2) + (34.64 \times 4) + (15 \times 7) - R_B \times 8 = 0$

$8R_B = 20 + 69.26 + 105 \Rightarrow 8R_B = 194.26$

$R_B = 194.26 / 8 = \underline{24.2825 \text{ kN}}$

$R_A = \underline{59.64} - R_B$

$= 59.64 - 24.2825$

$= \underline{\underline{35.3575 \text{ kN}}}$

b)

$$a_1 = 100 \times 50 = 5 \times 10^3 \text{ mm}^2$$

$$y_1 = 100 + 50/2 = 125 \text{ mm}$$

$$a_2 = 100 \times 50 = 5 \times 10^3 \text{ mm}^2$$

$$y_2 = 100/2 = 50 \text{ mm}$$

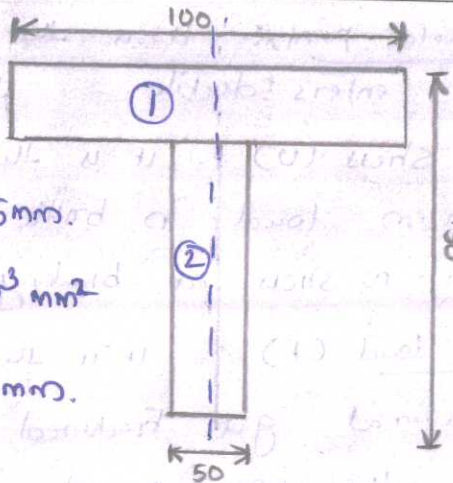
$$\bar{y} = \frac{a_1 y_1 + a_2 y_2}{a_1 + a_2}$$

$$= \frac{(5000 \times 125) + (5000 \times 50)}{5000 + 5000}$$

$$= \underline{\underline{87.5 \text{ mm}}}$$

The section is symmetrical about yy axis hence.

$$\bar{x} = 100/2 = \underline{\underline{50 \text{ mm}}}$$



9

15

IV

a) $R_A + R_D = (5 \times 2) + 18$

$$R_A + R_D = 10 + 18 = 28$$

$$\Sigma M_A = 0$$

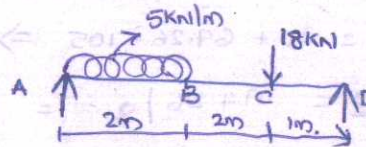
$$R_D \times 5 = (5 \times 2 \times 2/2) + (18 \times 4)$$

$$5R_D = 10 + 72$$

$$R_D = 82/5 = \underline{\underline{16.4 \text{ KN}}}$$

$$R_A = 28 - R_D$$

$$= 28 - 16.4 = \underline{\underline{11.6 \text{ KN}}}$$



6

$$b) a_1 = 100 \times 20 = 2000 \text{ mm}^2$$

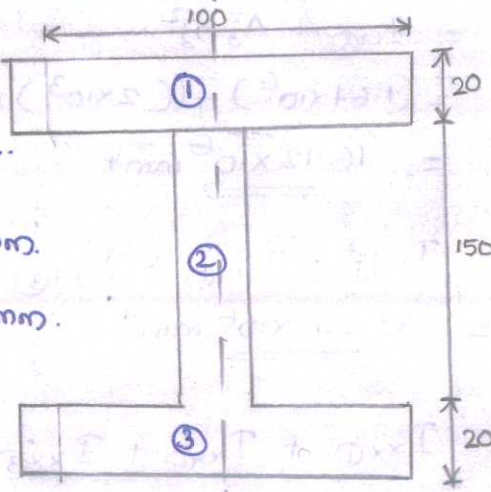
$$a_2 = 150 \times 20 = 3000 \text{ mm}^2$$

$$a_3 = 100 \times 20 = 2000 \text{ mm}^2$$

$$y_1 = 170 + 20/2 = 180 \text{ mm}$$

$$y_2 = 20 + 150/2 = 95 \text{ mm}$$

$$y_3 = 20/2 = 10 \text{ mm}$$



$$\bar{y} = \frac{a_1 y_1 + a_2 y_2 + a_3 y_3}{a_1 + a_2 + a_3} = \bar{y}$$

$$= \frac{(2000 \times 180) + (3000 \times 95) + (2000 \times 10)}{2000 + 3000 + 2000}$$

$$= \underline{95 \text{ mm}}$$

The Section is symmetrical about \bar{y} axis.

$$\bar{x} = 100/2 = \underline{50 \text{ mm}}$$

Moment of Inertia about YY -axis.

$$I_{u(1)} = \frac{db^3}{12} = \frac{20 \times 100^3}{12} = \underline{166666.67 \text{ mm}^4}$$

$$I_{u(2)} = \frac{db^3}{12} = \frac{150 \times 20^3}{12} = \underline{100 \times 10^3 \text{ mm}^4}$$

$$I_{u(3)} = \frac{db^3}{12} = \frac{20 \times 100^3}{12} = \underline{1.67 \times 10^6 \text{ mm}^4}$$

$$\begin{aligned} I_{YY(1)} &= I_{u(1)} + A_1 h_1^2 \\ &= (1.67 \times 10^6) + ((2 \times 10^3) \times (180 - 95)^2) \\ &= \underline{16.12 \times 10^6 \text{ mm}^4} \end{aligned}$$

$$\begin{aligned} I_{YY(2)} &= I_{u(2)} + A_2 h_2^2 \\ &= (1 \times 10^6) + ((3 \times 10^3) \times (95 - 95)^2) \\ &= \underline{1 \times 10^6 \text{ mm}^4} \end{aligned}$$

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4

$$\begin{aligned}
 I_{yy③} &= I_{c③} + A_3 h_3^2 \\
 &= (1.67 \times 10^6) + ((2 \times 10^3) \times (95-10)^2) \\
 &= \underline{\underline{16.12 \times 10^6 \text{ mm}^4}}
 \end{aligned}$$

$$\begin{aligned}
 I_{yy} &= I_{yy①} + I_{yy②} + I_{yy③} \\
 &= \underline{\underline{33.24 \times 10^6 \text{ mm}^4}}
 \end{aligned}$$

$$\begin{aligned}
 I_{xx} &= I_{xx①} + I_{xx②} + I_{xx③} \\
 &= \frac{bd^3}{12} + \frac{b_2 d_2^3}{12} + \frac{b_3 d_3^3}{12} \\
 &= \left(\frac{100 \times 20^3}{12} \right) + \left(\frac{40 \times 150^3}{12} \right) + \left(\frac{100 \times 20^3}{12} \right) \\
 &= \underline{\underline{5.7583 \times 10^6 \text{ mm}^4}}
 \end{aligned}$$

V

$$\text{a) i) } U = \frac{\sigma^2}{2E} \times V$$

$$\begin{aligned}
 \sigma &= \frac{P}{A} \\
 &= \frac{200 \times 10^3}{(500 \times 200)} \\
 &= 2 \text{ N/mm}^2
 \end{aligned}$$

$$\begin{aligned}
 P &= 200 \text{ kN} = 200 \times 10^3 \\
 b &= 500 \text{ mm} \\
 d &= 200 \text{ mm} \\
 L &= 2 \text{ m} = 2000 \text{ mm}
 \end{aligned}$$

$$= \frac{2^2}{2 \times 200 \times 10^3} \times (500 \times 200 \times 2000)$$

$$= \underline{\underline{2000 \text{ Nmm}}}$$

$$\text{ii) } U^* = \frac{\sigma^*^2}{2E} \times V$$

$$= \frac{200^2}{(2 \times 10^3) \times 2} \times (500 \times 200 \times 2000)$$

$$= 20 \times 10^6 \text{ Nmm}$$

$$\text{iii) modulus of Resilience} = \frac{\sigma^*^2}{2E}$$

$$= \frac{200^2}{2 \times 200 \times 10^3}$$

$$= \underline{\underline{0.1 \text{ N/mm}^2}}$$

9

b) Mechanical Properties of metals.

1) Stiffness (K) :- Ability of a material to resist deformation when it is stressed or it can also be defined as load required to create unit deformation in any structure.

2) Elasticity :- Property of a material to regain its original shape after deformation when external force is withdrawn.

3) Plasticity :- Property by which material retains the deformation after the withdrawal of load.

4) Ductility :- Property by which a metal can be drawn into wires.

5) Malleability :- Property by which a material can be drawn into thin sheet.

6) Brittleness :- the property by which material breaks with little permanent deformation.

7) Toughness :- Property of a material to resist fracture due to impact loads.

8) Hardness :- Property by which a material resist penetration or abrasion.

6 15

VI. a)

i) Resilience :- strain energy stored in a member when strained within elastic limits is known as Resilience.

$$U = \frac{\sigma^2}{2E} \times V$$

ii) Proof Resilience :- The maximum strain energy stored in a member when strained within elastic limits is known as proof Resilience.

$$U_p = \frac{\sigma_p^2}{2E} \times V$$

9

iii) Modulus of Resilience: It is defined as the proof resilience of a material per unit volume.

$$\text{Modulus of Resilience} = \frac{\text{Proof Resilience}}{\text{Volume}}$$

$$= \frac{\sigma^2}{2E}$$

b.

$$E = 200 \text{ GPa} \\ = 200 \times 10^9 \text{ N/m}^2$$

$$\sigma = \frac{2P}{A}$$

$$= \frac{2 \times 30 \times 10^3}{10^{-3}}$$

$$= 60 \times 10^6 \text{ N/m}^2$$

$$L = 2.5 \text{ m}$$

$$A = 1000 \text{ mm}^2 \\ = 10^{-3} \text{ m}^2$$

$$P = 30 \text{ kN} \\ = 30 \times 10^3 \text{ N}$$

$$U = \frac{\sigma^2}{2E} \times V$$

$$= \frac{(60 \times 10^6)^2}{2 \times (200 \times 10^9)} \times (2.5 \times 10^{-3})$$

$$= \frac{3600 \times 10^{12}}{400 \times 10^9} \times 2.5 \times 10^{-3}$$

$$= 22.5 \text{ Nm}$$

$$V = \pi r^2 h \\ = (10^{-3}) \times 2.5$$

$$= 2.5 \times 10^{-3}$$

$$\frac{U}{V} = \frac{\sigma^2}{2E}$$

6

15

VII a) $\Sigma V = 0$
 $R_A + R_B = 2 + (1 \times 3)$
 $R_A + R_B = 5 \quad \text{--- (1)}$

$\Sigma M_B = 0$

$4R_A = (1 \times 3 \times 1.5) + (2 \times 5)$

$R_A = 14.5 / 4$
 $= \underline{\underline{3.625 \text{ kN}}}$

$R_B = \underline{\underline{1.375 \text{ kN}}}$

SFD

$SF_D = -2 \text{ kN}, SF_A = 1.625 \text{ kN}$

$SF_C = 1.625 \text{ kN}, SF_B = -1.375 \text{ kN}$

BMD

$M_D = 0, M_A = -2 \text{ kNm}$

$M_C = -0.375 \text{ kNm}, M_B = 0$

$M_{\text{max}} = 0.94 \text{ kNm}$

Point of contraflexure

when $x = 0$

$M_{xx} = 0.375 + 1.625x - \frac{x^2}{2} = 0$

$x = \frac{-b \pm \sqrt{b^2 - 4ac}}{2a}$

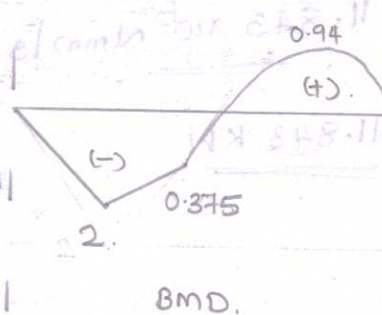
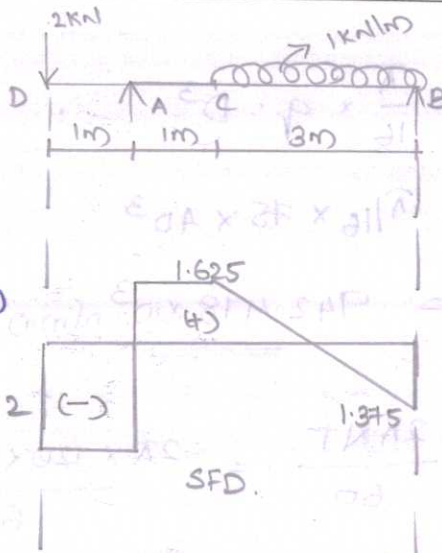
$a \rightarrow 0.5x^2$

$b \rightarrow 1.625x$

$c \rightarrow -0.375$

$x = \underline{\underline{3 \text{ m}}}$

$x = \underline{\underline{0.25 \text{ m}}}$



b.

$$T = \frac{\pi}{16} \times q \cdot D^3$$

$$= \frac{\pi}{16} \times 75 \times 40^3$$

$$= \underline{\underline{942.478 \times 10^3 \text{ Nmm}}}$$

$D = 40 \text{ mm}$
 $N = 120 \text{ rpm}$
 $q = 75 \text{ N/mm}^2$

$$P = \frac{2\pi NT}{60} = \frac{2\pi \times 120 \times 942.478 \times 10^3}{60}$$

$$\underline{\underline{11.843 \times 10^6 \text{ Nmm/s}}}$$

$$= \underline{\underline{11.843 \text{ kW}}}$$

5 15

VIII a)

$$\sum V = 0$$

$$R_A + R_B = 4 + 10 + 1$$

$$R_A + R_B = 15$$

$$\sum M_A = 0 \Rightarrow 5R_D =$$

$$((2 \times 2) \times 1 + (10 \times 4) + (1 \times 1) \times 5)$$

$$5R_D = 49.5$$

$$R_D = 49.5 / 5 = \underline{\underline{9.9 \text{ kN}}}$$

$$R_A = 5.1 \text{ kN}$$

SFD

$$SF_A = 5.1 \text{ kN}, SF_B = 1.1 \text{ kN}$$

$$SF_C = 1.1 \text{ kN}, SF_D = 1 \text{ kN}$$

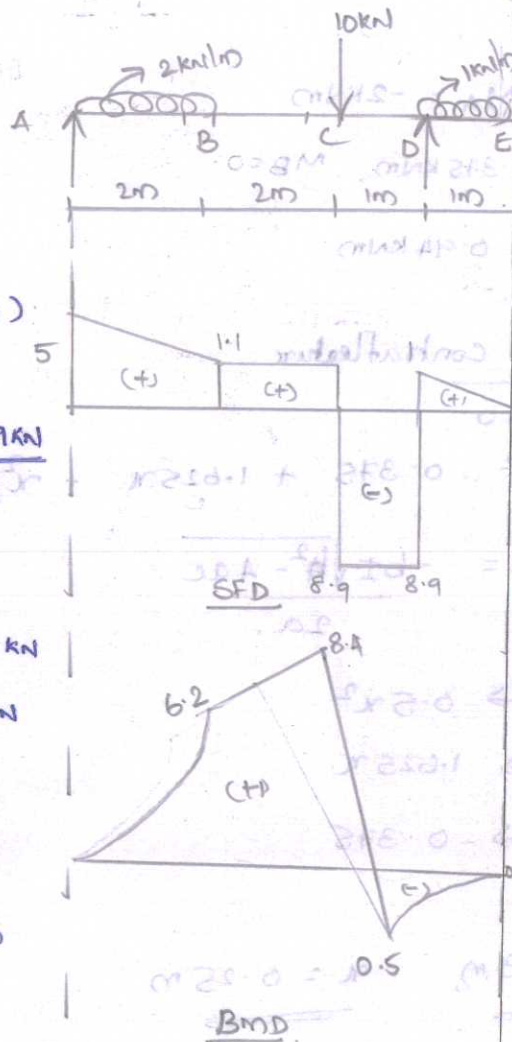
$$SF_E = 0$$

BMD

$$M_A = 0, M_B = 6.2 \text{ kNm}$$

$$M_C = 8.4 \text{ kNm}, M_D = -0.5 \text{ kNm}$$

$$M_E = 0$$



10

Point of Contraflexure

$$Max = 8.4 - 8.9x = 0$$

$$\Rightarrow 8.9x = 8.4 = x = \frac{8.4}{8.9} = 0.943 \text{ m}$$

b)

$$T = \frac{\pi}{16} \cdot z \left(\frac{D_o^4 - D_i^4}{D_o} \right)$$

$N = 200 \text{ rpm}$

$P = 300 \text{ kW}$

$z = 70 \text{ N/mm}^2$

$$P = \frac{2\pi N T}{60}$$

$$T = \frac{P \times 60}{2\pi N} = \frac{300 \times 10^3 \times 60}{2\pi \times 200}$$

$$= \underline{\underline{14323.945 \text{ Nm}}}$$

$$\frac{T \times 16}{\pi \times z} = \frac{D_o^4 - D_i^4}{D_o}$$

$$\frac{14.323 \times 10^6 \times 16}{\pi \times 70} = \frac{120^4 - D_i^4}{120}$$

~~$$D_i^4 = 120^4 - \frac{14.323 \times 10^6 \times 16 \times 120}{\pi \times 70}$$~~

~~$$D_i^4 = 207360000 - 61457443.29$$~~

$$120^4 - D_i^4 = 125051040$$

$$D_i^4 = 120^4 - 125051040$$

$$= 82308960.01$$

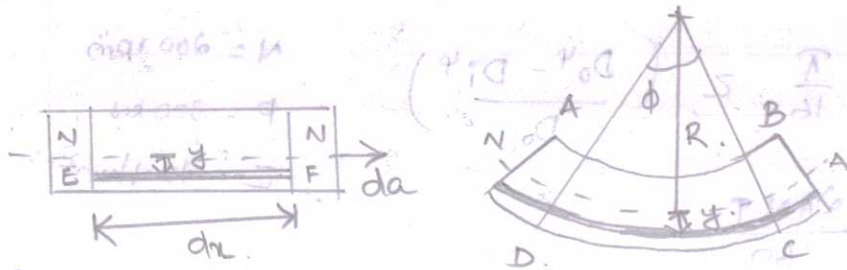
$$D_i = \underline{\underline{95.25 \text{ mm}}}$$

(10x)

5

15

1X a)



Consider a small length dl of beam subjected to simple bending.

$R \rightarrow$ Radius of neutral layer.

$\phi \rightarrow$ Angle subtended by $AB \& CD$ at centre.

Original length $EF = dl$.

$N'N = NN' = EF = dl$.

Strain in layers $EF = \frac{\text{changing length}}{\text{original length}}$

$$= \frac{E'F' - EF}{EF}$$

$$= \frac{(R+y)\phi - R\phi}{R\phi}$$

$$= \frac{y}{R} \quad \text{--- (1)}$$

Strain in layers $EF = \frac{\text{stress}}{E} = \frac{\sigma_b}{E} \quad \text{--- (2)}$

From (1) & (2)

$$\frac{\sigma_b}{E} = \frac{y}{R}$$

$$\boxed{\frac{\sigma_b}{y} = \frac{E}{R}} \quad \text{--- (3)}$$

Force in layers EF

$$\text{Force} = \text{stress} \times \text{Area}$$

$$= \sigma_b \times da$$

$$F = \left(\frac{E}{R} \times y \right) \times da$$

moment about Neutral axis.

Moment = Force \times lever arm

$$M = \left(\frac{E}{R} \times y \right) da \times y$$

$$M = \frac{E}{R} (y^2 da)$$

We know $I = y^2 da$

$$M = \frac{E}{R} \times I$$

$$\boxed{\frac{M}{I} = \frac{E}{R}} \quad \text{--- (4)}$$

From eqn (3) & (4) we can write bending eqn as,

$$\boxed{\frac{M}{I} = \frac{\sigma_b}{y} = \frac{E}{R}}$$

where,

$M \rightarrow$ Bending moment

$I \rightarrow$ Moment of Inertia

$\sigma_b \rightarrow$ bending stress.

$E \rightarrow$ young's modulus.

$R \rightarrow$ Radius of curvature.

$y \rightarrow$ Distance of fibre from the Neutral axis.

b) circumferential strain

$$E = 200 \text{ GPa}$$

$$e_c = \frac{Pd}{2tE} \left(1 - \frac{\mu}{2}\right)$$

$$d = 3000$$

$$t = 20 \text{ mm}$$

$$P = 3 \text{ N/mm}^2$$

$$\mu = 0.3$$

$$= \frac{3 \times 3000}{2 \times 20 \times 200 \times 10^3} \left(1 - \frac{0.3}{2}\right)$$

$$= \underline{\underline{9.5625 \times 10^{-4}}}$$

$$e_c = \frac{\Delta d}{d}$$

$$\text{change in dia} = \Delta d = e_c \times d$$

$$= 9.5625 \times 10^{-4} \times 3000$$

$$= \underline{\underline{2.868 \text{ mm}}}$$

Volumetric strain $e_v = \frac{Pd}{2tE} \left(\frac{5}{2} - 2\mu\right)$

$$= \frac{3 \times 3000}{2 \times 20 \times 200 \times 10^3} \times \left(\frac{5}{2} - 2 \times 0.3\right)$$

$$= \underline{\underline{2.1375 \times 10^{-3}}}$$

$$e_v = \frac{\Delta V}{V}$$

$$V = \frac{4}{3} \pi r^3$$

$$= 942477.96$$

$$\Delta V = e_v \times V$$

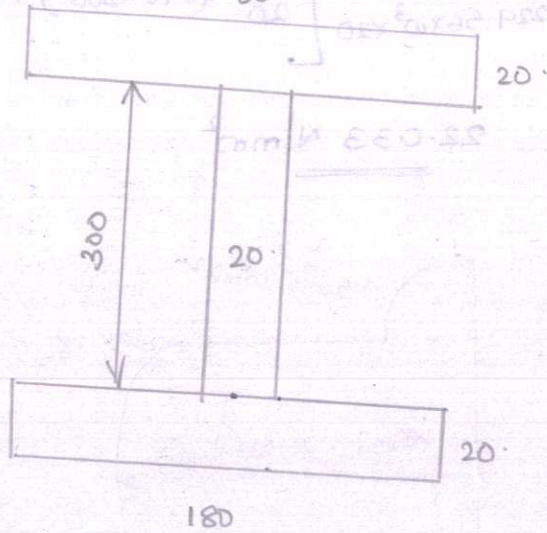
$$= 2.1375 \times 10^{-3} \times 942477.96$$

$$= \underline{\underline{20145.46 \text{ mm}^3}}$$

- X a) i) The material of the beam is homogenous & isotropic
- ii) The value of young's modulus of elasticity is the same in tension and compression.
- iii) The transverse sections which were plane before bending remains plane after bending
- iv) The beam is initially straight & longitudinal filaments bend into circular arc with a common centre of curvature.
- v) The radius of curvature is large compared with dimensions of the cross-section.

6

b)



15

$$\begin{aligned}
 I_{xx} &= \frac{BD^3}{12} - \frac{bd^3}{12} \\
 &= \left(\frac{180 \times 340^3}{12} \right) - \left(\frac{160 \times 300^3}{12} \right) \\
 &= \underline{\underline{229.56 \times 10^6 \text{ mm}^4}}
 \end{aligned}$$

9

