

1) (15)



GOVT. OF KERALA
DEPARTMENT OF TECHNICAL EDUCATION
OFFICE OF THE CONTROLLER OF TECHNICAL EXAMINATIONS
THIRUVANANTHAPURAM

DIPLOMA EXAMINATION IN ENGINEERING / TECHNOLOGY / MANAGEMENT

CORRECTION NOTE

Revision & Sub code: TED(15) - 3013
Subject: THEORY OF STRUCTURES - I
Verified by:

PART - B

This can also be considered.

II

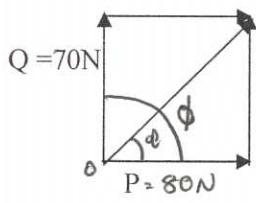
Resultant, $R = \sqrt{P^2 + Q^2}$

$$= \sqrt{70^2 + 80^2} = 106.3 \text{ N}$$

$$\phi = \tan^{-1} \frac{Q}{P}$$

$$= \tan^{-1} \frac{70}{80} = 0.875 = 41.18^\circ$$

Scheme of valuation
(scoring indicator)

Revision Course code TED (15) – 3013				
Course title THEORY OF STRUCTURES – I				
Q. No	Scoring indicator	Split up	Sub total	Total
I	<u>Part A</u>			
(1)	An effort which produces or tends to produce, destroys or tends to destroy motion	2	2	
(2)	The ratio between ultimate stress and working stress OR factor of safety = ultimate stress / working stress	2	2	
(3)	a layer in which neither tension nor compression	2	2	
(4)	Point load, UDL, Gradually varying load	2	2	
(5)	When beam subjected to BM, internal forces or stresses are developed in the section above and below NA, which are in opposite in nature. This form a couple and the moment couple is known as MOMENT OF RESISTANCE	2	2	10
II	Part B			
1.	<p style="text-align: center;">(Maximum marks : 30)</p> <div style="text-align: center;">  </div> $R = \sqrt{P^2 + Q^2 + 2PQ\cos\theta}$ $R = \sqrt{70^2 + 80^2 + 2 \times 70 \times 80 \times \cos 90} = 106.3 \text{ N}$	1		
		2		
		1		



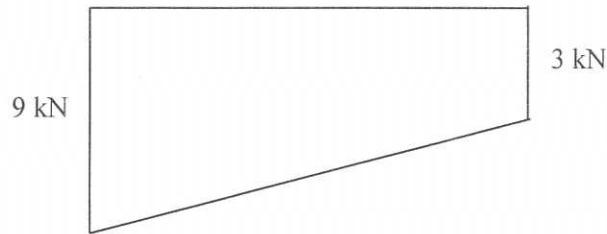
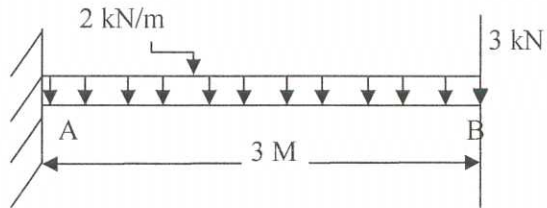
5.

SF at A = 3kN

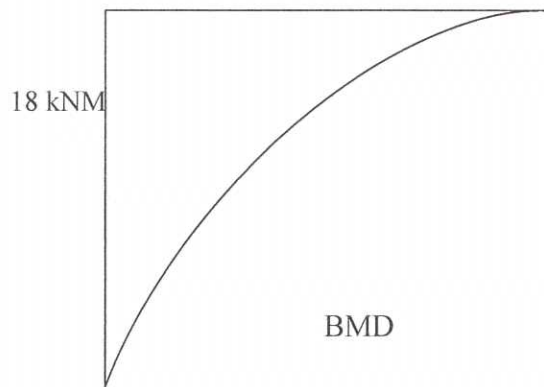
SF at B = 3+2*3=9kN

BM at ~~A~~^B = 0

BM at ~~B~~^A = 3*3+2*3*3/2=18 kNm



SFD



BMD

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1 (88)

1 (88)

6.

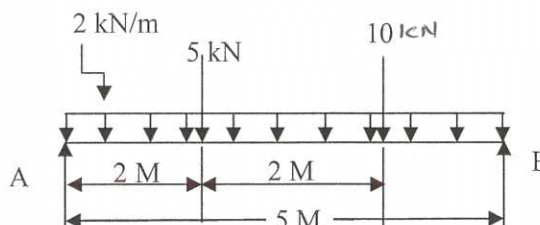
$$POWER, P = \frac{2\pi NT}{60}$$

$$105 \times 10^3 = \frac{2\pi 160 T}{60}$$

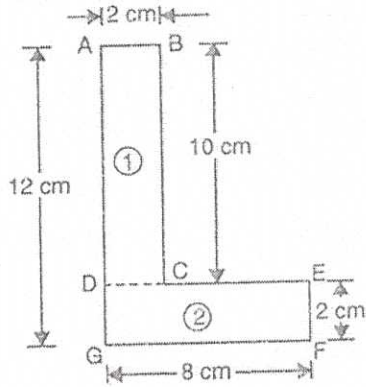
$$T = 6266.73 Nm$$

$$From\ shear\ stress\ consideration\ T = \frac{\pi f_s D^3}{16}$$

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<p>7.</p>	$6266.73 \times 1000 = \frac{\pi 65 D^3}{16}$ $D = 78.89 \text{ mm}$ <p>From stiffness consideration, $\frac{T}{J} = \frac{C\theta}{L}$</p> $\frac{6266.73 \times 1000}{\frac{\pi D^4}{32}} = \frac{8 \times 10^4 \times \pi}{3.5 \times 1000 \times 180}$ $D = 112.5 \text{ mm}$ <ol style="list-style-type: none"> 1. Material is homogeneous and isotropic. 2. Material is within elastic limit, Obeys Hooke's law. 3. The transverse section plane before bending remain plane after bending. 4. Each layer of the beam is free to expand or contract independently of the layer above or below it. 5. E is same in tension and compression. 	<p>1 1 1</p>	<p>6</p>	<p>30</p>
<p>III (a)</p>	<p style="text-align: center;">PART - C (Maximum marks : 60) (Answer one full question from each unit. Each full question carries 15 marks)</p> <p style="text-align: center;">UNIT-1</p>  <p>Error! Reference source not found. $R_A + R_B = 5 + 10 + 2 \times 5 = 25 \text{ kN}$</p> $5R_B = 5 \times 2 + 10 \times 4 + 2 \times 5 \times 2.5 = 75$ $R_B = 15 \text{ kN}$ $R_A = 10 \text{ kN}$	<p>1 (fig) 2 2</p>	<p>7</p>	

(b)



$$A_1 = 8 \times 2 = 16 \text{ cm}^2$$

$$A_2 = 10 \times 2 = 20 \text{ cm}^2$$

$$X_1 = 4 \text{ cm}$$

$$X_2 = 1 \text{ cm}$$

$$y_1 = 1 \text{ cm}$$

$$y_2 = 5 + 2 = 7 \text{ cm}$$

CG about y-axis

$$\bar{X} = \frac{A_1 X_1 + A_2 X_2}{A_1 + A_2}$$

$$\bar{X} = \frac{16 \times 4 + 20 \times 7}{16 + 20} = 2.33$$

CG about x-axis

$$\bar{Y} = \frac{A_1 Y_1 + A_2 Y_2}{A_1 + A_2}$$

$$\bar{Y} = \frac{8 \times 2 \times 1 + 10 \times 2 \times 7}{8 \times 2 + 10 \times 1} = 4.33 \text{ cm}$$

IV

$$(a) \quad \frac{P}{\sin \alpha} = \frac{Q}{\sin \beta} = \frac{R}{\sin \gamma}$$

$$\frac{15}{\sin 90} = \frac{T_1}{\sin 150} = \frac{T_2}{\sin 120}$$

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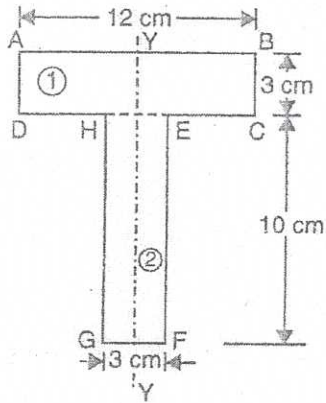
2

$T_1 = 7.5N$

$T_2 = 12.99N$

OR

(b)



CG ABOUT X AXIS

$A_1 = 3 \times 10 = 30\text{cm}^2, Y_1 = 5\text{cm}$

$A_2 = 3 \times 12 = 36\text{cm}^2, Y_2 = 10 + 1.5 = 11.5\text{cm}$

$$\bar{Y} = \frac{A_1 Y_1 + A_2 Y_2}{A_1 + A_2}$$

$$\bar{Y} = \frac{30 \times 5 + 36 \times 11.5}{30 + 36} = 8.55\text{cm}$$

$$I_{xx1} = \frac{b_1 d_1^3}{12} + A_1 h_1^2$$

$$I_{xx1} = \frac{3 \times 10^3}{12} + 10 \times 3 \times 3.55^2 = 628.07\text{cm}^4$$

$$I_{xx2} = \frac{b_2 d_2^3}{12} + A_2 h_2^2$$

$$I_{xx2} = \frac{12 \times 3^3}{12} + 12 \times 3 \times 2.95^2 = 340.29\text{cm}^4$$

$$I_{xx} = I_{xx1} + I_{xx2} = 628.07 + 340.29 = 968.36\text{cm}^4$$

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9

UNIT-II				
V				
(a)	$P = 300kN, A_s = 5600mm^2, \frac{E_s}{E_c} = 15$ $A = 250 \times 250 = 62500mm^2$ $A_c = 62500 - 5600 = 56900mm^2$ $\frac{P_c}{E_c} = \frac{P_s}{E_s}$ $P_s = \frac{P_c}{E_c} E_s = 15P_c$ $P = P_s A_s + P_c A_c$ $300000 = 5600 \times 15P_c + 56900P_c$ $P_c = 2.14N/mm^2$ $P_s = 15 \times 2.14 = 31.94N/mm^2$	$\frac{1}{2}$ $\frac{1}{2}$ 1 1 1 1 1 1 1	7	
(b)	<ol style="list-style-type: none"> 1. Strength :-capacity of materials to withstanding the breaking, bowing or deforming under the action of load. 2. Elasticity :-property of a material to come back to its original size and shape after removal of load. 3. Plasticity:-property of a material that makes it to be in the deformed size and shape even after the withdrawal of load. 4. Ductility:-property of a material that allows it to deform or make into thin wires under the action of tensile load. 5. Malleability:-property in which a thin sheet can be easily formed without breaking by hammering or rolling or pressing. 6. Stiffness:-it is the resistance of an elastic body to deflection or deformation by the applied load 7. Toughness:-ability of a material to absorb energy and plastic deformation without fracture. Or energy per unit volume that a material can be absorb before rupture. 8. Brittleness:-material subjected to stress, it breaks without significant 	1 1 1 1 1 1 1		

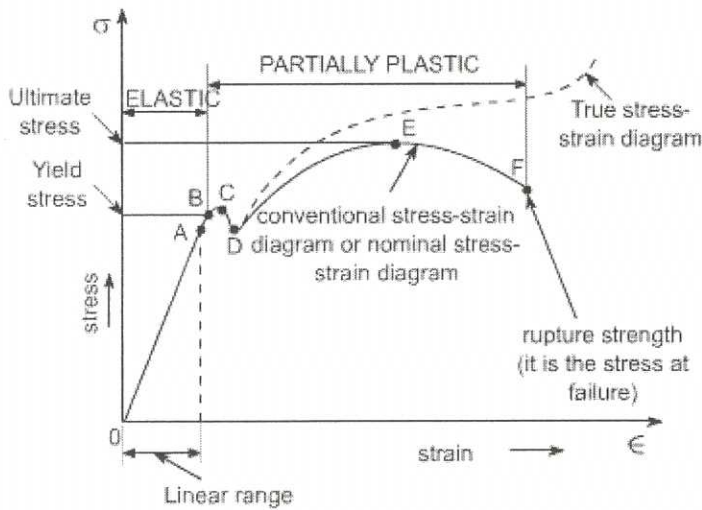
plastic deformation. It is the property of a material that fractures when subjected to stress but has a little tendency to deform before rupture.

9. Hardness:-the property of a material to resist pressing or scratch of a sharp object.

(any eight)

OR

VI
(a)



SALIENT POINTS OF THE GRAPH:

(A) So it is evident from the graph that the strain is proportional to strain or elongation is proportional to the load giving a st.line relationship. This law of proportionality is valid upto a point A. Or we can say that point A is some ultimate point when the linear nature of the graph ceases or there is a deviation from the linear nature. This point is known as **the limit of proportionality or the proportionality limit**.

(B) For a short period beyond the point A, the material may still be elastic in the sense that the deformations are completely recovered when the load is removed. The limiting point B is termed as **Elastic Limit**.

(C) and (D) - Beyond the elastic limit plastic deformation occurs and strains are not totally recoverable. There will be thus permanent deformation or permanent set when load is removed. These two points are termed as upper and lower yield

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2 (fr)

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	<p>points respectively. The stress at the yield point is called the yield strength.</p> <p>(E) A further increase in the load will cause marked deformation in the whole volume of the metal. The maximum load which the specimen can with stand without failure is called the load at the ultimate strength.</p> <p>The highest point 'E' of the diagram corresponds to the ultimate strength of a material.</p> <p>(F) Beyond point E, the bar begins to forms neck. The load falling from the maximum until fracture occurs at F</p>	1		
(b)	$l = 4m = 4000mm$ $b = 35mm$ $d = 20mm$ $P = 30kN = 30000N$ $\frac{1}{m} = 0.25$ $E = 200GPa = 200000N/mm^2$ $A = 35 \times 20 = 700mm^2$ $dl = \frac{Pl}{AE} = \frac{30000 \times 4000}{700 \times 200000} = 0.857mm$ $linearstrain = \frac{\delta l}{l} = \frac{0.857}{4000} = 0.00021$ $lateralstrain = \frac{1}{m} linearstrain = 0.25 \times 0.00021 = 0.000054$ $volumeV = 4000 \times 35 \times 20 = 2800000$ $\frac{dv}{V} = \frac{P}{bdE} (1 - \frac{2}{m}) = \frac{30000}{35 \times 20 \times 200000} (1 - 2 \times 0.25)$ $= 1.07 \times 10^{-4}$ $dv = 2800000 \times 1.07 \times 10^{-4} = 300mm^3$	1	1	8
VII	1. Simply Supported Beam			
(a)	<p>when the ends of a beam are formed to stand freely on supports beam, it is known as a simple (freely) supported beam. It contains pinned support at one end and roller support at the other end.</p>	1	1	7

2. Fixed Beam

If a beam is inflexible at both ends in order that the slope at the ends become zero, it is known as fixed beam. It is also called a built-in beam. The fixed ends produce fixing moments there other than the reactions.

3. Cantilever Beam

If a beam is fixed at one end whereas the other end is free, it is termed as cantilever beam.

4. Continuously Supported Beam

If a beam supports more than two supports are arranged to the beam, it is described as continuously supported beam.

5. overhanging beam

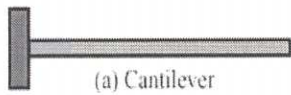
If the ends of the beams are project beyond the ends of the support.

6. Propped cantilever

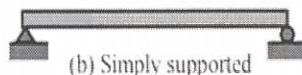
One end of the beam is fixed and other end is simply supported.

OR

Types of Beams



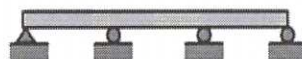
(a) Cantilever



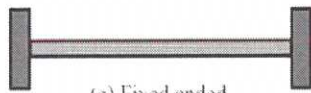
(b) Simply supported



(c) Overhanging



(d) continuous



(e) Fixed ended



(f) Cantilever, simply supported

Any five

(b) $R_A + R_D = 1 + 2 + 2 \times 10 = 23kN$
 $10R_D = 1 \times 2 + 2 \times 5 + 2 \times 10 \times 5$
 $R_D = 11.2kN$
 $R_A = 11.8kN$

5x1 5

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<p>Shear force</p> <p>$F_D = 11.2kN$ $F_C = 11.2 - 2 \times 5 - 2 = 0.8kN$</p> <p>$F_B = 11.2 - 2 \times 8 - 2 - 1 = 7.8kN$ $F_A = 11.8kN$</p> <p>Bending moment</p> <p>$M_A = M_D = 0$ $M_C = 11.2 \times 5 - 2 \times 5 \times 2.5 = 31.0kNm$ $M_B = 11.2 \times 8 - 2 \times 8 \times 4 - 2 \times 3 = 19.6kNm$</p> <p>OR</p> <p>VIII</p> <p>(a)</p> <p>$d = 30cm, \theta = 1.5^\circ = 1.5 \times \frac{\pi}{180} rad, L = 8m = 8000mm$</p> <p>$f_s = 40N/mm^2, C = 8 \times 10^4 N/mm^2$</p> <p>$J = \frac{\pi D^4}{32} = \frac{\pi 30^4}{32}$</p> <p>$\frac{T}{J} = \frac{C\theta}{L}$</p> <p>$\frac{T}{\frac{\pi 30^4}{32}} = \frac{8 \times 10^4 \times \frac{\pi}{180}}{8 \times 1000}$</p> <p>$T = 20818.69 Nmm$</p> <p>$T = \frac{\pi f_s D^3}{16} = \frac{\pi \times 40 \times 30^3}{16} = 212057.5 Nmm$</p> <p>$\therefore \text{Maximum torque } T = 20818.69 Nmm$</p>		<p>1</p> <p>1</p> <p>1</p> <p>1 (SFD)</p> <p>1 (BMD)</p> <p>1</p> <p>1</p> <p>10</p>		
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(b)

$$d = 530 - 2 \times 10 = 510 \text{ mm}$$

$$\text{circumferential} \dots \text{stress} = \sigma_1 = \frac{pd}{2t} = \frac{10.5 \times 510}{2 \times 10} = 267.75 \text{ N/mm}^2$$

$$\text{Longitudinal} \dots \text{stress} = \sigma_2 = \frac{\sigma_1}{2} = 133.875 \text{ N/mm}^2$$

$$\text{circumferential} \dots \text{strain} = \epsilon_1 = \frac{1}{E} (\sigma_1 - \frac{\sigma_2}{m})$$

$$\epsilon_1 = \frac{1}{2.1 \times 10^5} (267.75 - 133.875 \times 0.3) = 0.00108$$

$$\text{Increase in ex. dia} = \epsilon_1 \times D = 530 \times 0.00108 = 0.5745 \text{ mm}$$

$$\text{longitudinal} \dots \text{strain} = \epsilon_2 = \frac{1}{E} (\sigma_2 - \frac{\sigma_1}{m})$$

$$\epsilon_2 = \frac{1}{2.1 \times 10^5} (133.875 - 267.75 \times 0.3) = 0.000255$$

$$\text{Increase in length} = \epsilon_2 \times L = 1800 \times 0.000255 = 0.459 \text{ mm}$$

UNIT-IV

IX

M = bending moment at the beam

Θ = angle subtended at the centre of the arc

R = radius of curvature of the beam

$$\delta l = PQ - P^1Q^1$$

$$\text{Strain} = \epsilon = \frac{\delta l}{l} = \frac{PQ - P^1Q^1}{PQ}$$

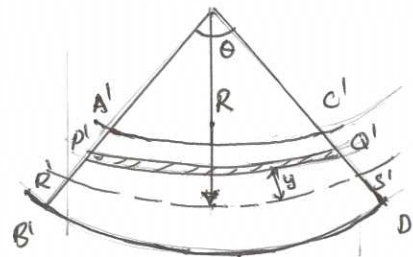
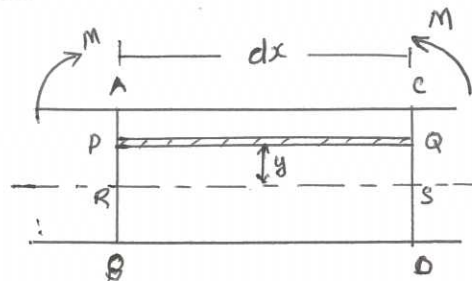
$$\text{From geometry } \frac{P^1Q^1}{R^1S^1} = \frac{R-y}{R}$$

$$1 - \frac{P^1Q^1}{R^1S^1} = 1 - \frac{R-y}{R}$$

$$\frac{R^1S^1 - P^1Q^1}{R^1S^1} = \frac{y}{R}$$

$$\therefore R^1S^1 = PQ$$

$$\frac{PQ - P^1Q^1}{PQ} = \frac{y}{R}$$



Bending of beam

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1 (Fig)

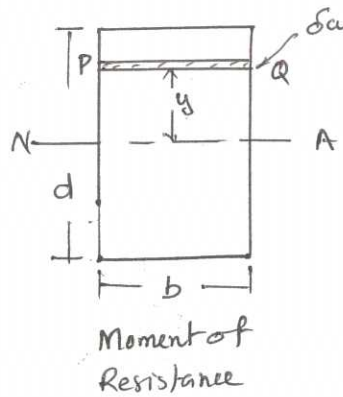
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$$\epsilon = \frac{y}{R}$$

$$\sigma = \epsilon \times E$$

$$\sigma = \frac{y}{R} \times E$$

$$\therefore \frac{\sigma}{y} = \frac{E}{R}$$



Let δa = area of layer PQ

Stress in layer PQ $\sigma = \frac{y}{R} \times E$

Force in layer PQ = $y \times \frac{E}{R} \delta a$

Moment of these force about NA = $y^2 \times \frac{E}{R} \delta a$

Algebraic sum of all such moments about NA $M = \sum \frac{E}{R} y^2 \delta a = \frac{E}{R} \sum y^2 \delta a$

$\sum y^2 \delta a$ represents the MI of whole section about NA

$$M = \frac{E}{R} I$$

$$\frac{M}{I} = \frac{E}{R} \quad \text{AND} \quad \frac{\sigma}{y} = \frac{E}{R}$$

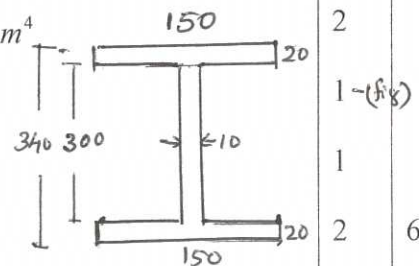
$$\frac{M}{I} = \frac{\sigma}{y} = \frac{E}{R}$$

(b) $SF = 50kN = 50000N$

$$I = \frac{B^3 D^3 - b^3 d}{12} = \frac{150 \times 340^3 - 140 \times 300^3}{12} = 1763000000 mm^4$$

$$A\bar{y} = 150 \times 20 \times 160 + 150 \times 10 \times 75 = 592500$$

$$q = \frac{F A \bar{y}}{I b} = \frac{50000 \times 592500}{176300000 \times 10} = 16.8 N / mm^2$$



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OR

X

$$I_{xx} = \frac{BD^3 - bd^3}{12}$$

(a)

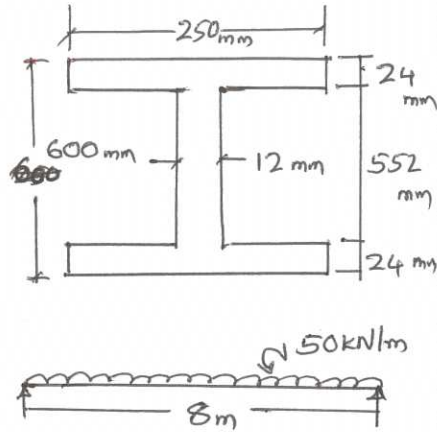
$$I_{xx} = \frac{250 \times 600^3 - 238 \times 552^3}{12} = 1164100 \text{ mm}^4$$

$$M = \frac{wl^2}{8} = \frac{50 \times 8^2}{8} = 400 \text{ kNm} = 400 \times 10^6$$

$$\frac{M}{I} = \frac{f}{y}$$

$$\frac{400 \times 10^6}{1164100} = \frac{f}{300}$$

$$f = 103.08 \text{ N/mm}^2$$



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1 (fig)

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(b)

$$f_s = 28 \text{ N/mm}^2, \tau = 2 \text{ N/mm}^2$$

$$I = \frac{bd^3}{12} = \frac{100 \times 150^3}{12} = 28125000 \text{ mm}^4$$

$$M = \frac{wl^2}{8} = \frac{w \times 10^6}{2} \text{ Nmm}$$

$$\frac{M}{I} = \frac{f_s}{y}$$

$$\frac{\frac{w \times 10^6}{2}}{28125000} = \frac{28}{75}$$

$$w = 21 \text{ kN/m}$$

$$SF = \frac{wl}{2} = \frac{w \times 2}{2} = w \text{ kN}$$

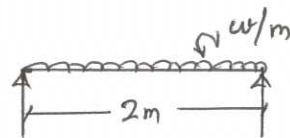
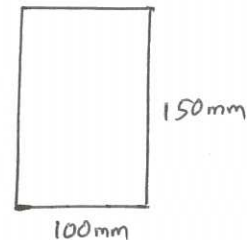
$$A\bar{y} = 100 \times 75 \times \frac{75}{2} = 281250$$

$$\tau = \frac{FA\bar{y}}{Ib}$$

$$2 = \frac{w \times 10^3 \times 281250}{28125000 \times 100}$$

$$w = 20 \text{ kN/m}$$

$$\therefore w = 20 \text{ kN/m}$$



1 (fig)

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